

## GB923238

Publication Title:

Improvements in or relating to air conditioning systems for aircraft

Abstract:

Abstract of GB923238

923,238. Ventilating aircraft &c. NORMA- LAIR Ltd. Oct. 31, 1960 [Oct. 30, 1959], No. 36858/59. Class 137. Fresh air from the engine intake 42 and pipe 41 and recirculated air from the cabin 7 and fan 10 are mixed and flow through cooler 1 and deodorizer 3 back to the cabin which has vents 30 opening automatically to give a maximum pressurization. Oxygen from supply 4 is supplied to the conditioned air downstream of deodorizer 3 when required. The cooler 1 is the evaporator of a compression refrigerating system 20, the condenser 24 of which is cooled by ram air. A cooler 40 in pipe 41 is also cooled by ram air. An ozonizer may be provided.

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# PATENT SPECIFICATION

923,238

DRAWINGS ATTACHED.

*Inventor:—PETER WILLIAM FITT.*



*Date of filing Complete Specification : Oct. 31, 1960.*

*Application Date : Oct. 30, 1959. No. 36858/59.*

*Complete Specification Published : April 10, 1963.*

*Index at Acceptance :—Class 137, B3F, B4(B:F:H).*

*International Classification :—F24f.*

## COMPLETE SPECIFICATION.

### Improvements in or relating to Air Conditioning Systems for Aircraft.

We, NORMALAIR LIMITED, of West Hendford, Yeovil, in the County of Somerset, a British Company, do hereby declare the invention, for which we pray that a patent  
5 may be granted to us, and the method by which it is to be performed, to be particularly described in and by the following statement:—

This invention relates to an air conditioning system for use particularly, although not  
10 exclusively in aircraft of the passenger carrying type, and more particularly, although not exclusively to such aircraft of the type that operate at high altitudes and at speeds  
15 considerably in excess of speed of sound.

Known methods of inducing air from the outside temperatures at a rate sufficient to conform with operation requirements, presents certain problems of cooling which  
20 cannot readily be overcome by normal means. In a passenger aircraft a considerable intake of oxygen would be required for say 150 passengers during a flight of 3 hours. As a result a considerable quantity  
25 of carbon dioxide will be produced as well as a substantial amount of water vapour. It is therefore of necessity for the reasons stated and further to avoid undesirable odours, to supply about 1 lb. of fresh air  
30 per person per minute for such passenger aircraft, even where there is a considerable degree of internal recirculation of the air.

It is with this problem in mind that the following invention has been evolved which  
35 has for its object to offset the adverse effects due to cooling by providing a system for reducing the air flow into the cabin and thereby to reduce the cooling necessary.

[Price 4s. 6d.]

It is a further object of the invention to  
40 minimise the problems associated with known means for supplying the passengers with air or other breathable gases at a suitable temperature and thereby reduce the penalty of aerodynamic drag and weight  
45 associated with such means.

Accordingly the invention consists in an air conditioning system for an enclosure of an aircraft comprising first means for supplying  
50 breathable gas to said enclosure and means for recirculating said gas within said enclosure to condition same, characterised in that second means for supplying breathable gas, other than ambient air, are provided for introducing gas into said enclosure  
55 in a manner as to augment said first breathable gas supply and to dilute carbon dioxide within said enclosure.

Further objects and advantages of the invention will become readily apparent from the following detailed description with reference  
60 to the accompanying diagrammatic drawing which shows a schematic drawing of the system according to the invention.

In carrying the invention into effect according to one convenient form by way  
65 of example only, referring to the accompanying diagrammatic drawing, we provide an evaporator indicated at 1, being in communication by way of conduit 2 with a deodoriser 3 employing activated charcoal.  
70 An oxygen cylinder, liquid oxygen converter or like oxygen source 4 conveniently provided outside or within the cabin space generally indicated at 7, having a flow controller (not shown) in the supply line, is in  
75 communication with conduit 5 disposed at

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the outlet of the deodoriser 3. The conduit 5 is provided with a plurality of outlets generally indicated at 6 for allowing breathable gas to be discharged to the cabin space 7. A motor driven recirculating fan indicated at 10 upstream of said evaporator 1 is provided for recirculating breathable gas from the cabin space to said evaporator 1 by way of conduit 11. A by-pass conduit 12 is provided in such a manner that some of the breathable gas for recirculation may by-pass the evaporator 1 as required. The evaporator 1 is in conduit communication with a typical aircraft cooling system for example of the compression, evaporation and condensation type generally indicated at 20, forming no part of the invention, which comprises a compressor 21 driven by an electric motor 22 in communication with an evaporator 1 by way of conduit 23. The compressor is in communication with a condenser 24 by way of conduit 25 and the outlet of said condenser 24 is in communication with said evaporator 1 by way of conduits 26, 27 and expansion valve 28. A plurality of discharge valves generally indicated at 30 are provided in the region of the aircraft within the toilet and galley space generally indicated at 31. Air is bled from the engine air-intake 42 to conduit 11 disposed upstream of said evaporator 1 by way of heat exchanger 40 and conduit 41.

The cooling system utilises a liquid refrigerant such as dichlorotetra-fluoroethane, hereinafter referred to as the refrigerant, which in operation, leaves the compressor 21 driven by electric motor 22, in a gaseous state at a substantially high temperature and pressure and flows to condenser 24 by way of conduit 25. The refrigerant liquifies and cools, heat of liquefaction being transferred to the ram air drawn over condenser 24 by way of convenient means. The refrigerant then passes to expansion valve 28 where it partially vaporises cooling to a substantially low temperature with a corresponding reduction in pressure. The refrigerant on leaving expansion valve 28, flows to evaporator 1 where it is entirely converted to gas and thereby air from the engine air intake 42 and recirculation air from cabin space 7 passing through evaporator 1 by way of conduit 11 is cooled by evaporation of the refrigerant within evaporator 1. The refrigerant is finally returned to compressor 21 by way of conduit 23.

In operation of the system according to the invention, external air is supplied from the engine air intake 42 at relatively high temperature to evaporator 1 by way of conduits 41 and 11, the temperature thereof being substantially reduced by heat exchanger 40, where heat is given up to ram air passing over said heat exchanger 40.

The air is further reduced in temperature

and substantially dried by the action of the evaporator 1 due to the operation of the cooling system 20 as hereinbefore described. The air then passes by way of conduit 2 to deodoriser 3, where any obnoxious odours are removed therefrom. The air is then discharged into the cabin space 7 by way of outlets 6 in conduit 5.

Upon discharge into the cabin space 7, the air is used for normal breathing purposes and it will be realised that air normally exhaled will contain carbon dioxide and water vapour. Also air within the cabin space 7 may contain any odours which may originate from the passengers.

The air in the cabin space 7 is therefore recirculated to ensure an even distribution of conditioned air therein in part, by recirculation over the evaporator 1 by way of a fan 10 and conduit 11, where any water vapour is removed. The air is then passed by way of conduit 2 to deodoriser 3 where any obnoxious odours contained therein are removed as hereinbefore described. The air is also adapted, in part to by-pass the evaporator 1 by way of by-pass 12 to prevent excess dehumidification.

If no effort is made to replace oxygen consumed by the passengers, the oxygen percentage will fall. Provided the selected equivalent operating altitude of the cabin is sufficiently low, the slight reduction in the oxygen percentage is acceptable, otherwise a metered flow of oxygen in the cabin space 7 is required equal to the average consumption of the occupants. In order to effect this and also to dilute the carbon dioxide within the cabin space 7, additional oxygen may be supplied from oxygen container 4 by way of a flow controller (not shown).

It will be apparent that due to air being continually supplied from the engine air intakes 42 by way of heat exchanger 40 to cabin space 7 and also oxygen being supplied from the oxygen container 4, pressure within the cabin space 7 will rise, and to control this pressure, a plurality of discharge valves 30 are provided. The discharge valves 30 are adapted to open at a predetermined pressure and allow air to be discharged from the cabin space 7 to the atmosphere. At operational altitudes the pressure within the cabin space 7 will always be in excess of the pressure of the surrounding atmosphere and therefore an outflow of air will always be experienced from the cabin. The discharge valves 30 are therefore positioned within the toilet and galley space 31 in order that the flow of air to atmosphere, may at the same time, remove a quantity of the obnoxious odours therein.

Often a small quantity of ozone is desirable to act as a disinfectant and although this exists in the outer atmosphere and is

obviously introduced to the cabin space 7  
by way of engine air intakes 42 it may be  
found that this is totally destroyed by the  
high temperatures in the engine. In this  
5 case it will be necessary to provide an  
ozoniser for this purpose which may com-  
prise electrical means operated by H.T.  
supply.

10 It will be appreciated and understood by  
those skilled in the art that numerous modi-  
fications and refinements may be carried out  
without departing from the scope of the in-  
vention and the system hereinbefore des-  
15 cribed has numerous applications and is not  
necessarily restricted to aircraft of the super-  
sonic high altitude passenger carrying type.

#### WHAT WE CLAIM IS:—

20 1. An air conditioning system for an  
enclosure of an aircraft, comprising means  
for supplying breathable gas to said en-  
closure and means for recirculating said gas  
within said enclosure to condition same,  
25 characterised in that supplementary means  
for supplying oxygen or breathable oxygen  
mixture, other than ambient air, are provided  
for introducing gas into said enclosure in  
a manner as to augment said breathable  
gas supply and to dilute carbon dioxide  
within said enclosure.

30 2. An air conditioning system as claimed  
in Claim 1, and including a plurality of dis-  
charge valves for discharging gas from said  
enclosure such that a predetermined pressure  
is maintained therein.

35 3. An air conditioning system as claimed  
in Claim 1, and further including heat ex-  
change means for cooling gas from said  
means for supplying breathable gas before  
discharge into said enclosure.

40 4. An air conditioning system as claimed  
in Claim 3, wherein said heat exchange  
means comprise a first heat exchanger  
cooled by air flow, and a second heat ex-  
45 changer comprising an evaporator of a re-  
frigeration system.

5. An air conditioning system as claimed  
in Claim 4, wherein said refrigeration system  
is of the closed circuit compression, con-  
densation and evaporation type.

50 6. An air conditioning system as claimed  
in Claim 4, wherein said gas from said  
means for supplying breathable gas after  
being cooled by said second heat exchanger,

is passed through de-odourising means, and  
gas from said supplementary means is intro- 55  
duced to augment said first supply before  
discharge into said enclosure.

7. An air conditioning system as claimed  
in Claim 4, wherein said second heat ex- 60  
changer has by-pass means associated there-  
with.

8. An air conditioning system as claimed  
in Claims 1—7, wherein said means for re- 65  
circulating said gas within said enclosure  
comprise a motor driven fan adapted to re-  
circulate gas from said enclosure and pass  
same in part over said second heat ex-  
changer thereby removing water vapour  
therefrom and in part by said by-pass  
means, said gas being de-odourised and 70  
augmented by way of gas from said supple-  
mentary means, before discharge to said  
enclosure.

9. An air conditioning system as claimed  
in Claim 8, wherein said recirculated gas so 75  
augmented dilutes any carbon dioxide with-  
in the enclosure in addition to maintaining  
the oxygen level therein.

10. An air conditioning system as  
claimed in Claim 2, wherein said discharge 80  
valves are disposed within toilet and galley  
space in such a manner that outflow of air  
from said space removes a substantial quan-  
tity of odours therefrom.

11. An air conditioning system as 85  
claimed in Claim 10, and including ozonis-  
ing means within said cabin space for dis-  
infectant purposes.

12. An air conditioning or like system  
for an enclosure of an aircraft, characterised 90  
by a cabin space having means for supply-  
ing breathable gas thereto, fan means for  
recirculating breathable gas therein, breath-  
able gas augmenting means such as a pres-  
sure bottle, liquid gas converter or the like, 95  
for diluting carbon dioxide therein, de-  
odourizing means and means for partially  
dehumidifying said breathable gas, ozonis-  
ing means for disinfectant purposes and a  
plurality of discharge valves for discharging 100  
excess gas from said cabin space.

13. An air conditioning system for an  
enclosure of an aircraft substantially as  
hereinbefore described with reference to the  
accompanying drawings. 105

For the Applicants,  
L. H. HAYWARD.

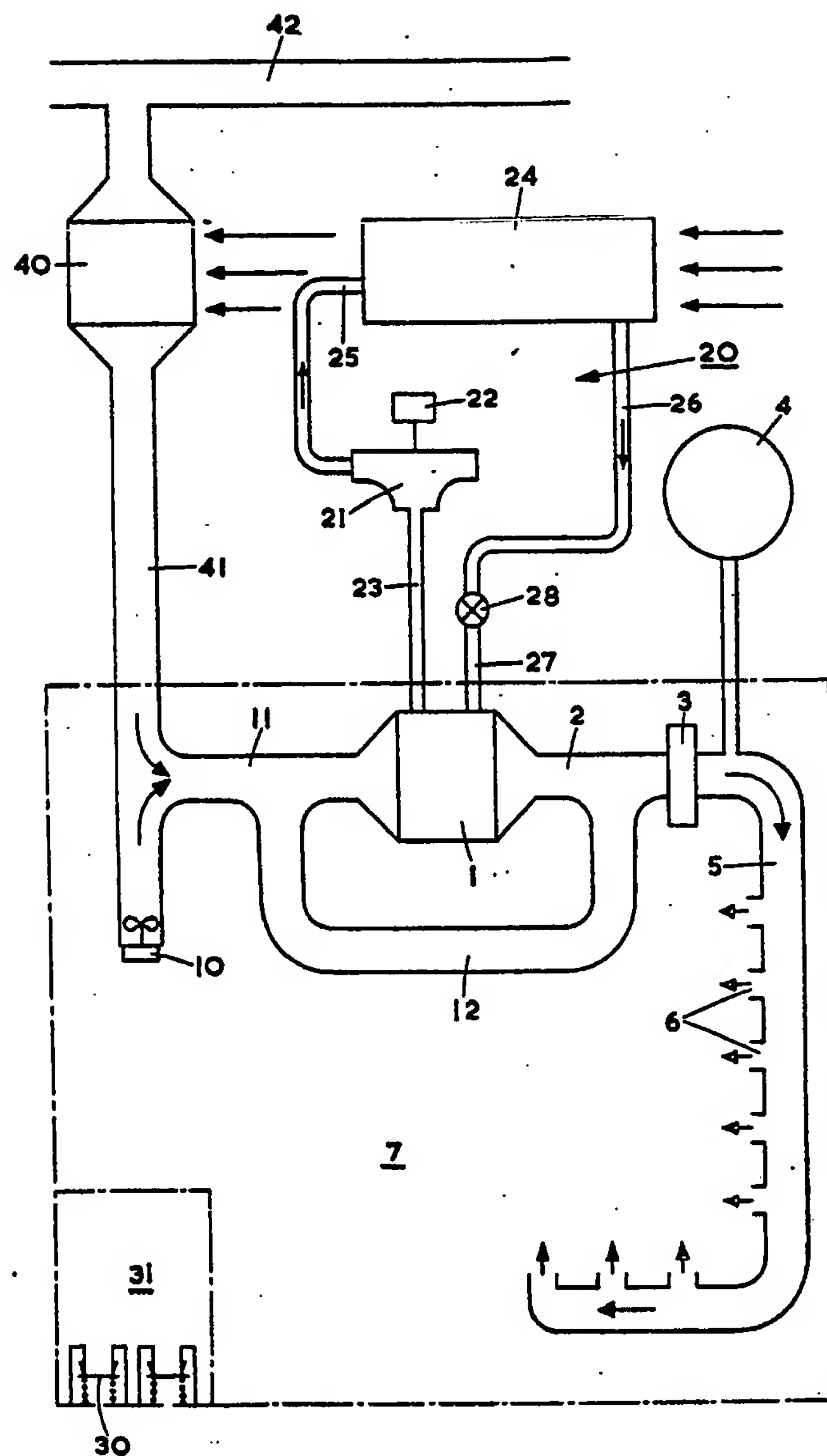


923238

COMPLETE SPECIFICATION

1 SHEET

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